



---

# **Heated Promoted Combustion- Initial Test Results**

**Dr. Carl D. Engel - Qualis Corporation**

**Stephen Herald - ICRC**

**S. Eddie Davis - Marshall Space Flight Center**

Marshall Space Flight Center  
Materials Combustion Research Center  
ICRC/Qualis Corporation



- 
- The Purpose of the STD 6001 test 17 is to determine the flammability of materials in GOX at ambient temperature and at use pressure.
  - The purpose of the new Heated Promoted combustion test is to determine the flammability of materials in GOX at use temperature and pressure.

**Objective: Present the new Heated Promoted Combustion method and show initial data and trends for three representative metals.**



## MSFC New Facility

---

# High-Temperature High-Pressure Promoted Ignition-Combustion Test

- The direct heating method at MSFC uses an **induction coil** housed inside the promoted ignition-combustion chamber, which is similar to the chamber described in ASTM 124-95.
- The MSFC improved induction heated high-pressure promoted ignition-combustion chamber can sustain a pressure of up to **10,000 psia** (6.9 MPa) in pure gaseous oxygen, and contains an induction coil capable of preheating a metal specimen to near its melting temperature [possibly in excess of **2,000° F** (1,100 C)] depending upon the test pressure.



## Heated Promoted Combustion Tester

---



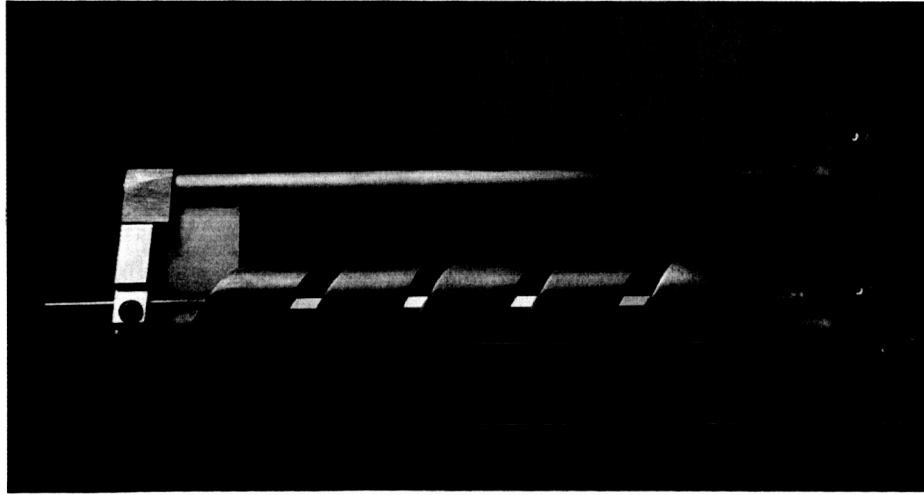
- Six site ports were arranged in a helical pattern to allow the observation of the combustion event for the entire length of the sample
- These site ports, due to the higher pressures involved in the test, are constructed out of 1.3-inch thick sapphire.
- Thermal imaging camera used for temperature measurement

Stephen Herald, aligning one of six cameras in the chamber view port.

Marshall Space Flight Center  
Materials Combustion Research Center  
ICRC/Qualis Corporation



## Improved Induction-Coil Design

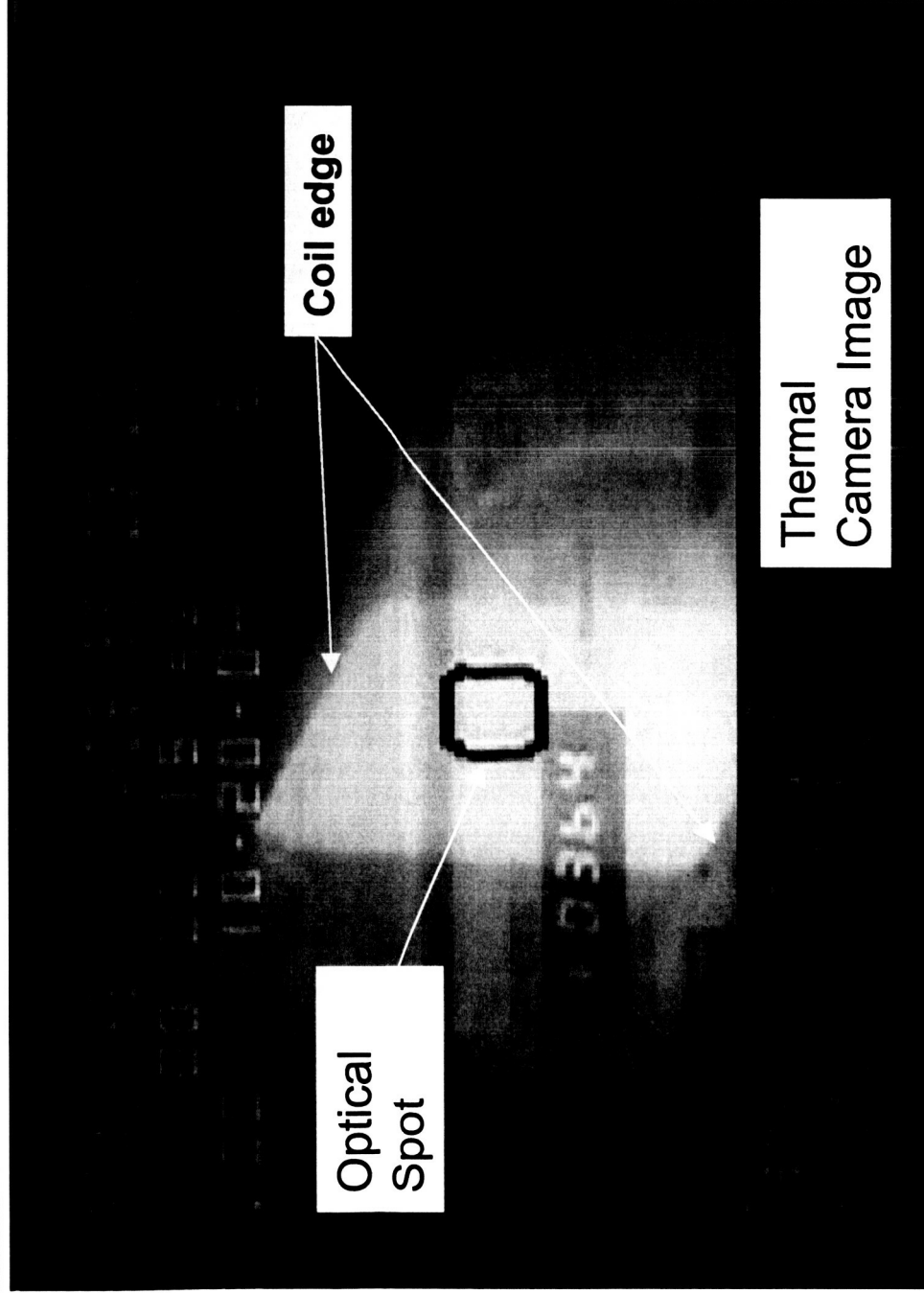


- An induction coil capable of preheating a metal specimen to near its melting temperature [possibly in excess of 2,000° F (1,100 C)] depending upon the test pressure.
- Viewing through spiral openings

Marshall Space Flight Center  
Materials Combustion Research Center  
ICRC/Qualis Corporation

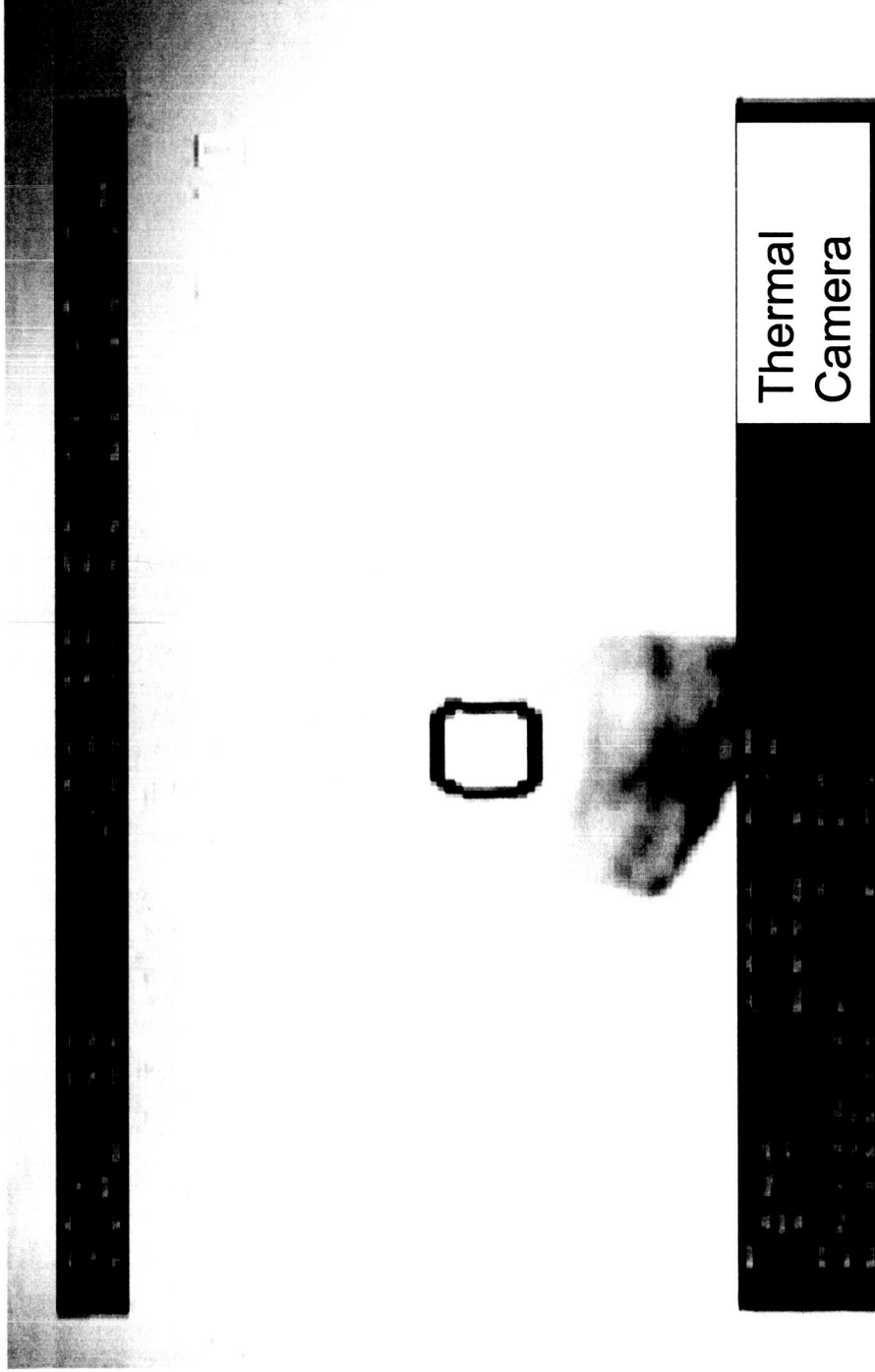


## Induction-Heated Rod (1689 °F)



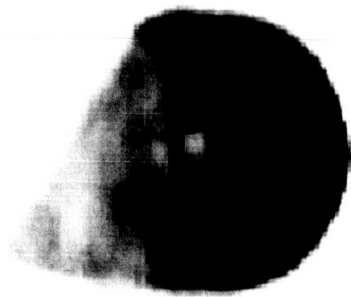
Marshall Space Flight Center  
Materials Combustion Research Center  
ICRC/Qualis Corporation

## Burning Tip Emerging (2292 °F)



Marshall Space Flight Center  
Materials Combustion Research Center  
ICRC/Qualis Corporation

# Droplet End Centered (3668 °F)



Thermal  
Camera

Marshall Space Flight Center  
Materials Combustion Research Center  
ICRC/Qualis Corporation



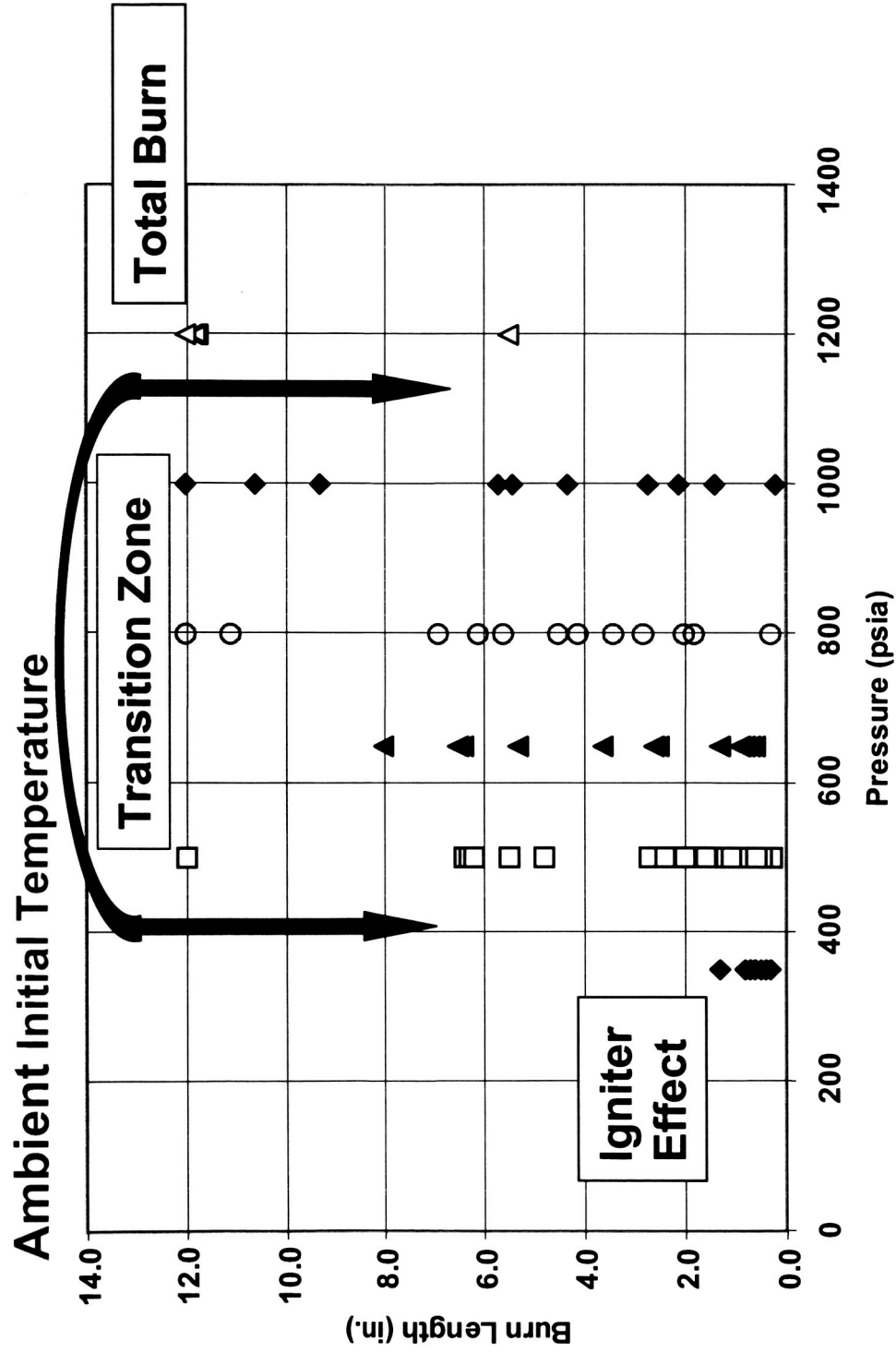
## **Outline**

---

- 1. Review standard test data at ambient average temperature**
- 2. Review Heated promoted combustion data at elevated temperatures and pressures**
- 3. Review Heated promoted combustion data burn rates at elevated temperatures and pressures**
- 4. Examine the implications of this data and testing technique on future vehicle design**



## Raw Promoted Combustion Burn Length Data (SS 347)

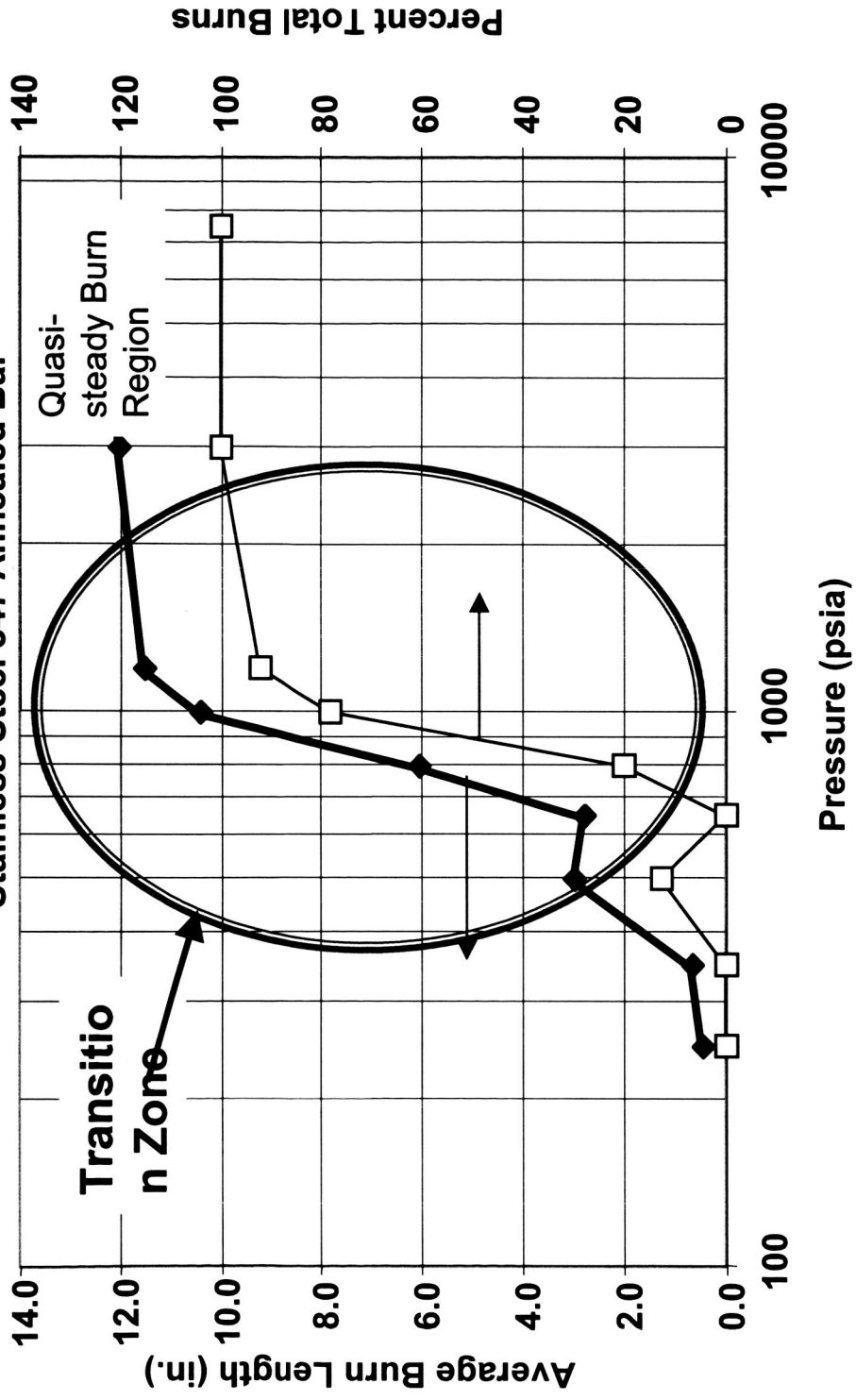




## SS 347 Total Burn Percent

### Ambient Initial Temperature

Stainless Steel 347 Annealed Bar





## Similar Observations by Other Investigators

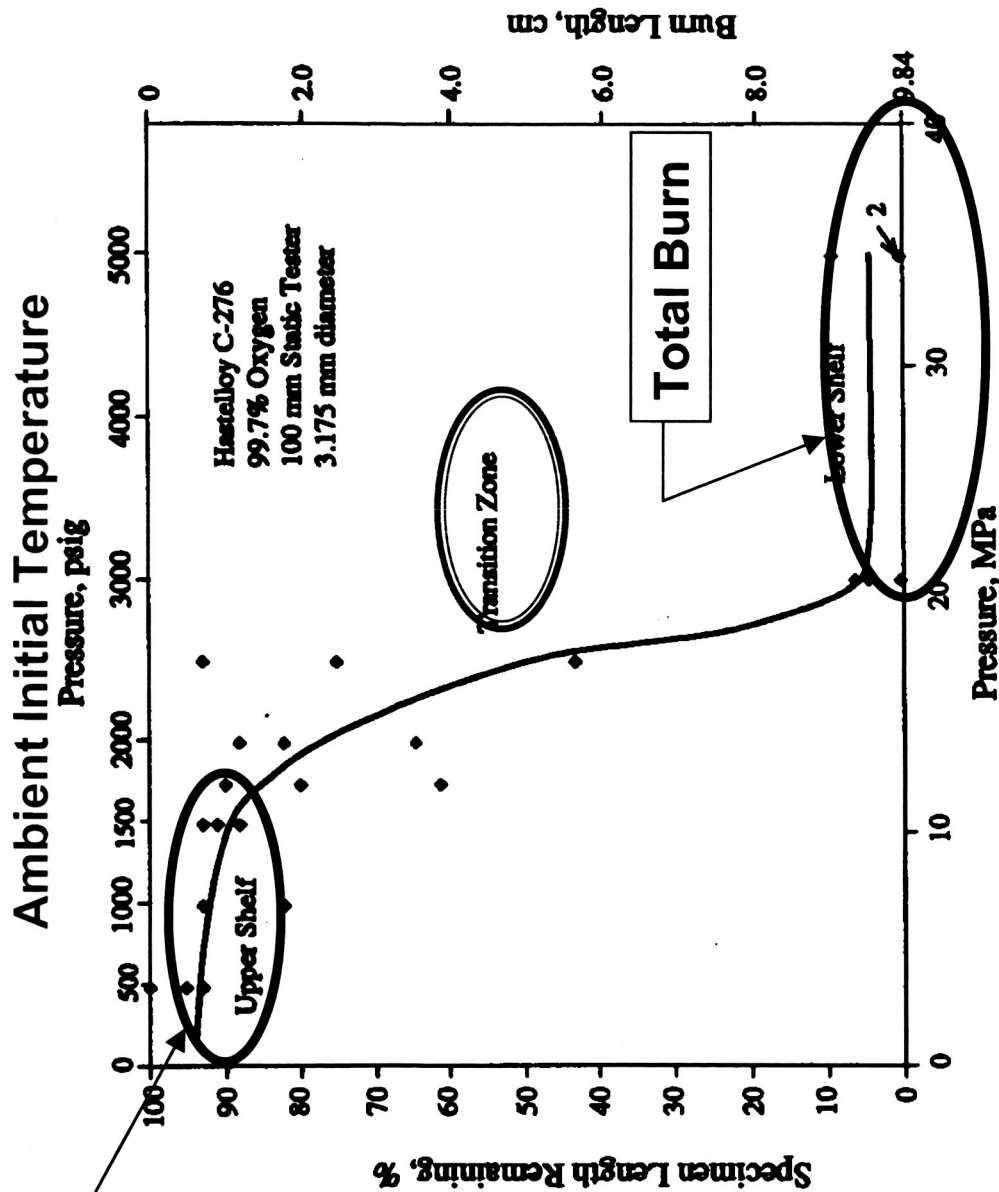
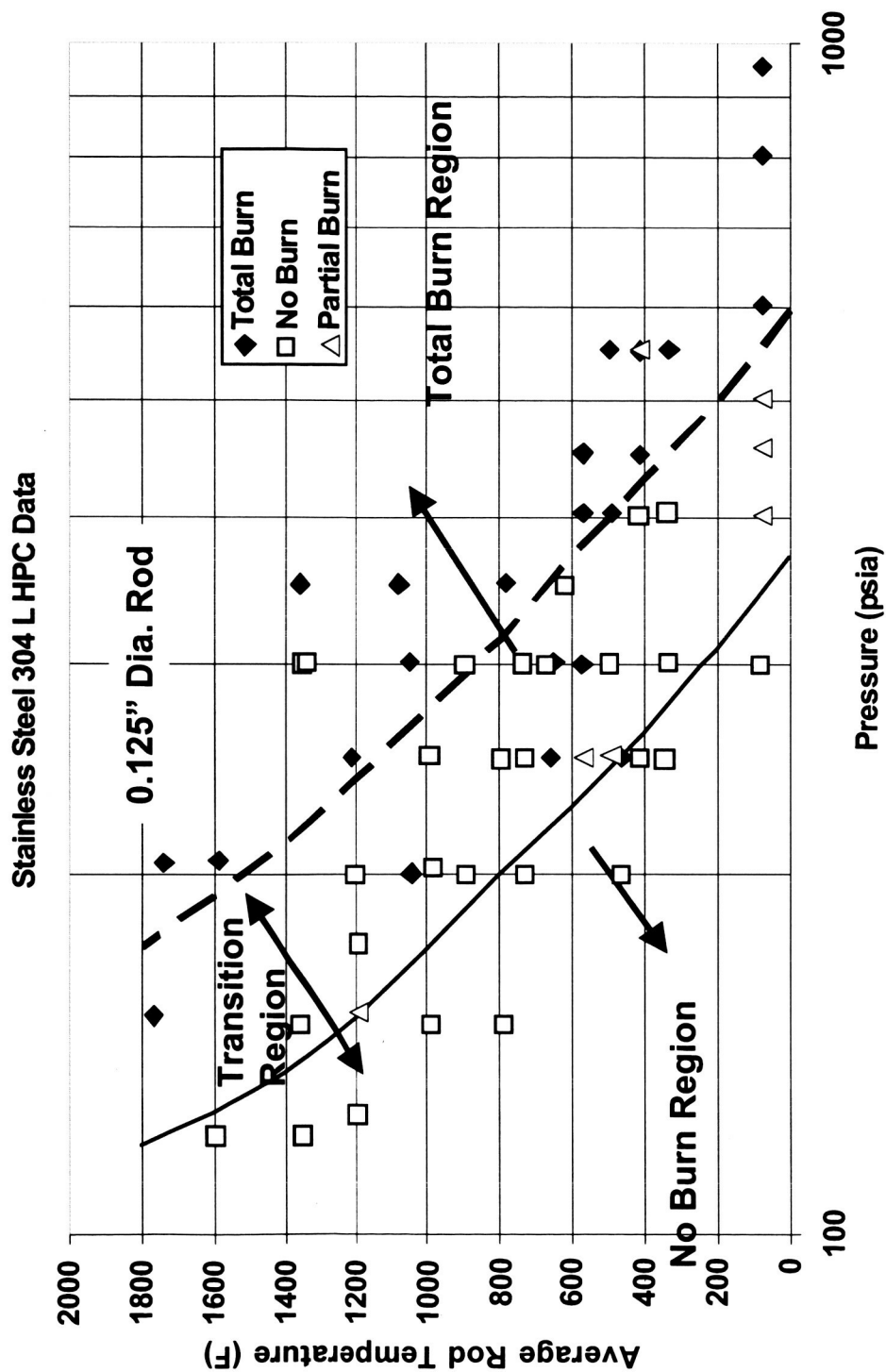


Fig. 5 from  
Zawierucha, R.,  
K. McIlroy & R.  
Mazzarella,  
ASTM Fifth Vol.,  
STP 1111, 1991





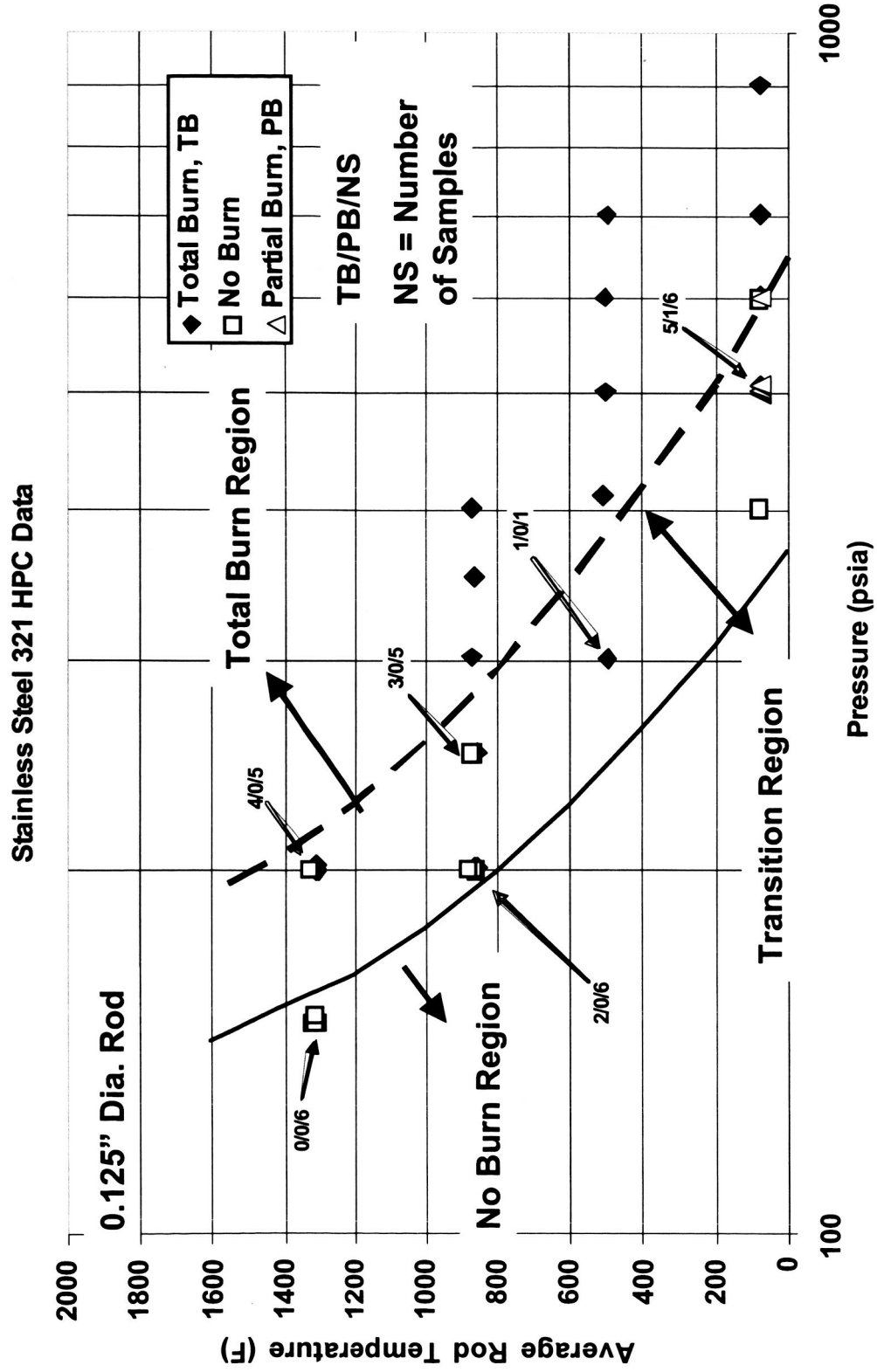
## SS 304L Pressure -Temperature Burn Characteristics



Marshall Space Flight Center  
Materials Combustion Research Center  
ICRC/Qualis Corporation



# SS 321 Pressure - Temperature Burn Characteristics

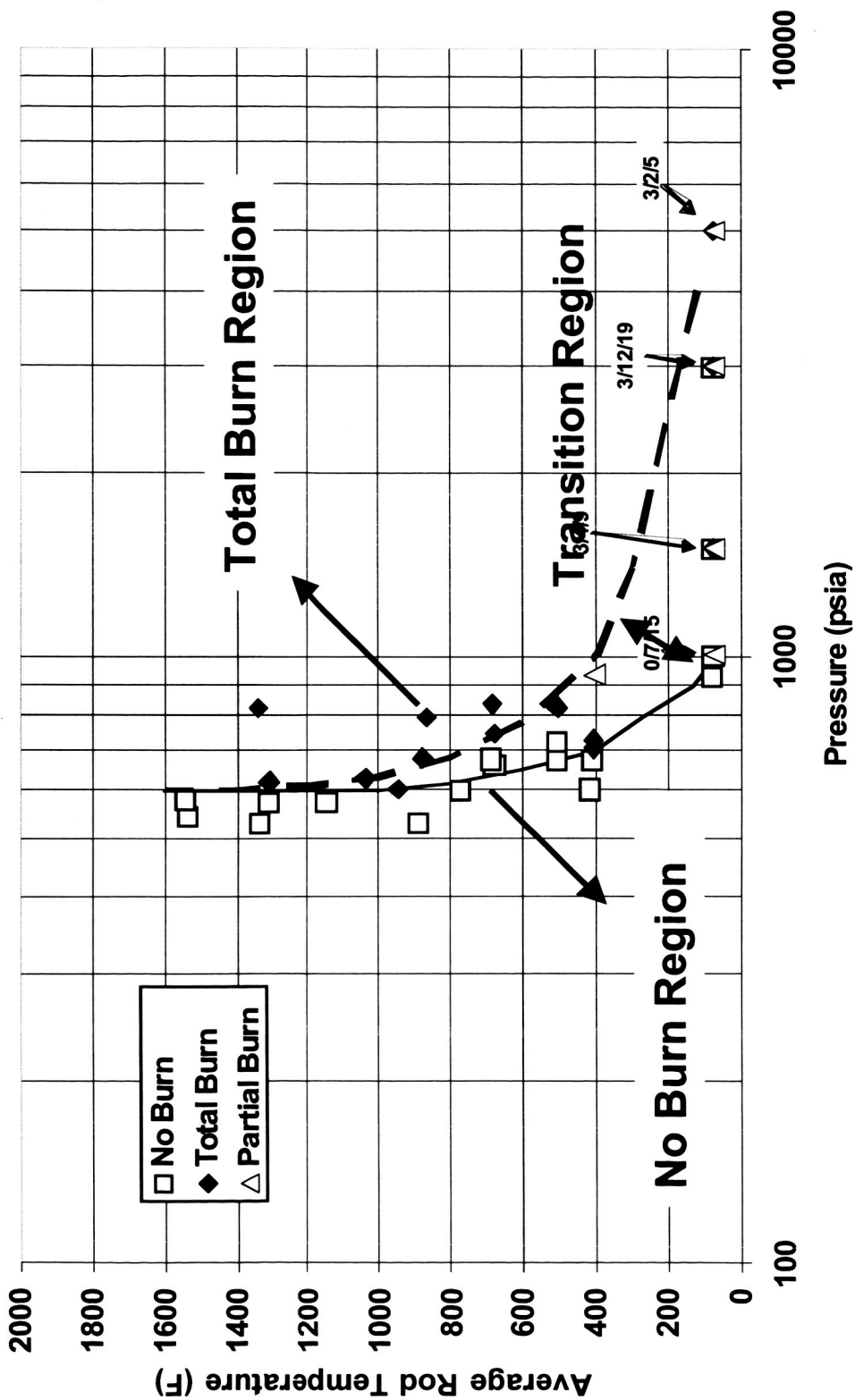




# Haynes 214 Pressure – Temperature Burn Characteristics

Haynes 214 HPC Data

Initial results

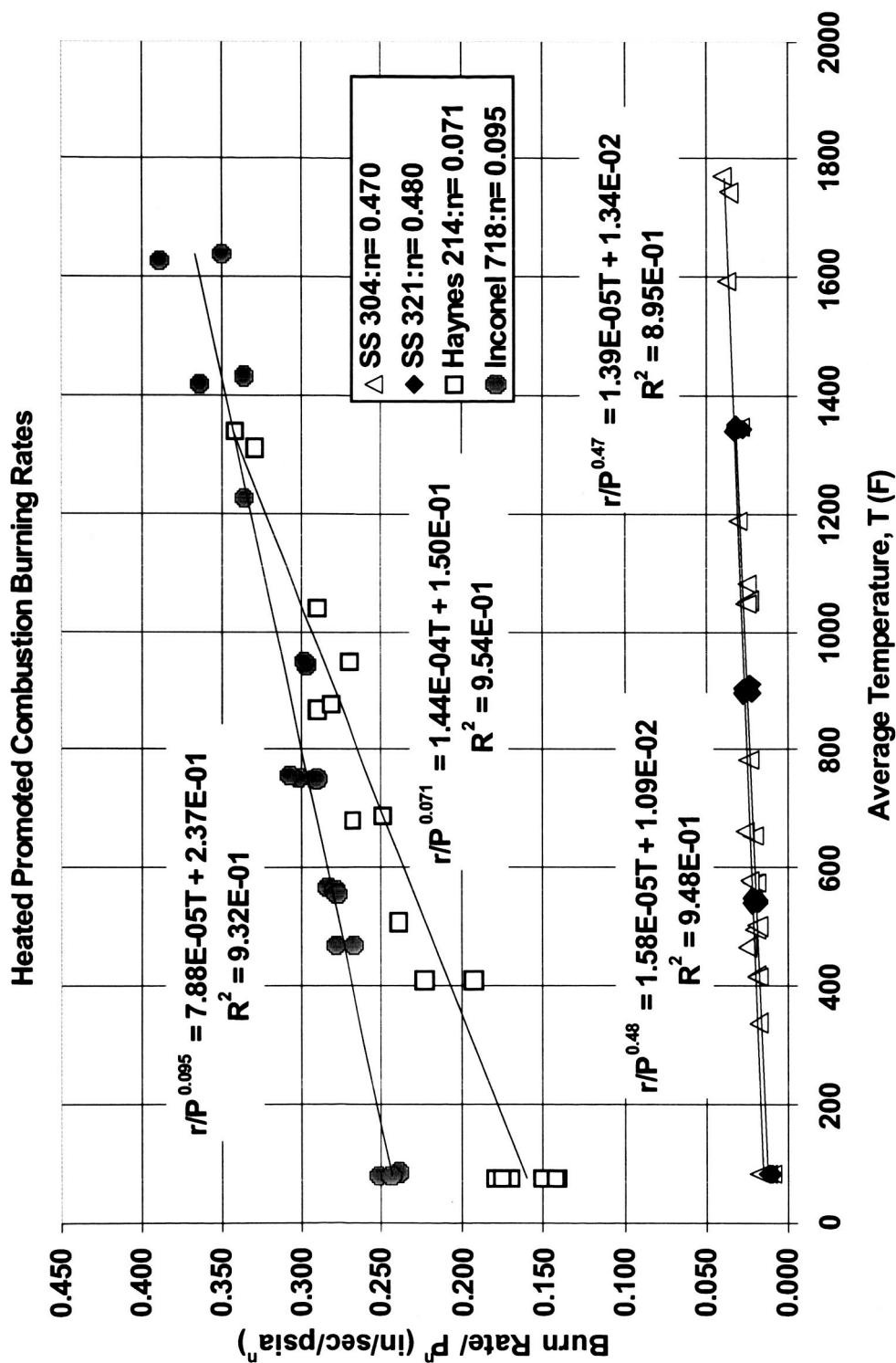


Marshall Space Flight Center  
Materials Combustion Research Center  
ICRC/Qualis Corporation





# Composite Burn Rates



Marshall Space Flight Center  
Materials Combustion Research Center  
IC RC/Qualis Corporation



## Observations

---

- **Elevated average temperature significantly reduces the Ignition Pressure of the tested Metals**
- **Four tested metals: Stainless Steel 304L, Haynes 214, Inconel 718, and Stainless Steel 321 all exhibit a very significant reduction in the pressure where Ignition and total burning occurs compared to ambient initial temperature conditions.**
- **The total burn rate data appear to correlate quite well with an exponential power of pressure and a linear function of average temperature.**
- **The elevated average temperature effect has significant implications as to the oxygen compatibility of metals at high metal working temperatures.**



## Implications

---

### Testing

**Material environmental condition ranges (P,T) must be defined and tested.**

- **A much larger set of runs must be made to determine ignition risks.**
- **A different testing protocol must be established.**
- **Engineering math models must be established to extend tested sample data to application conditions**